

Aircraft Configuration Design PART-I

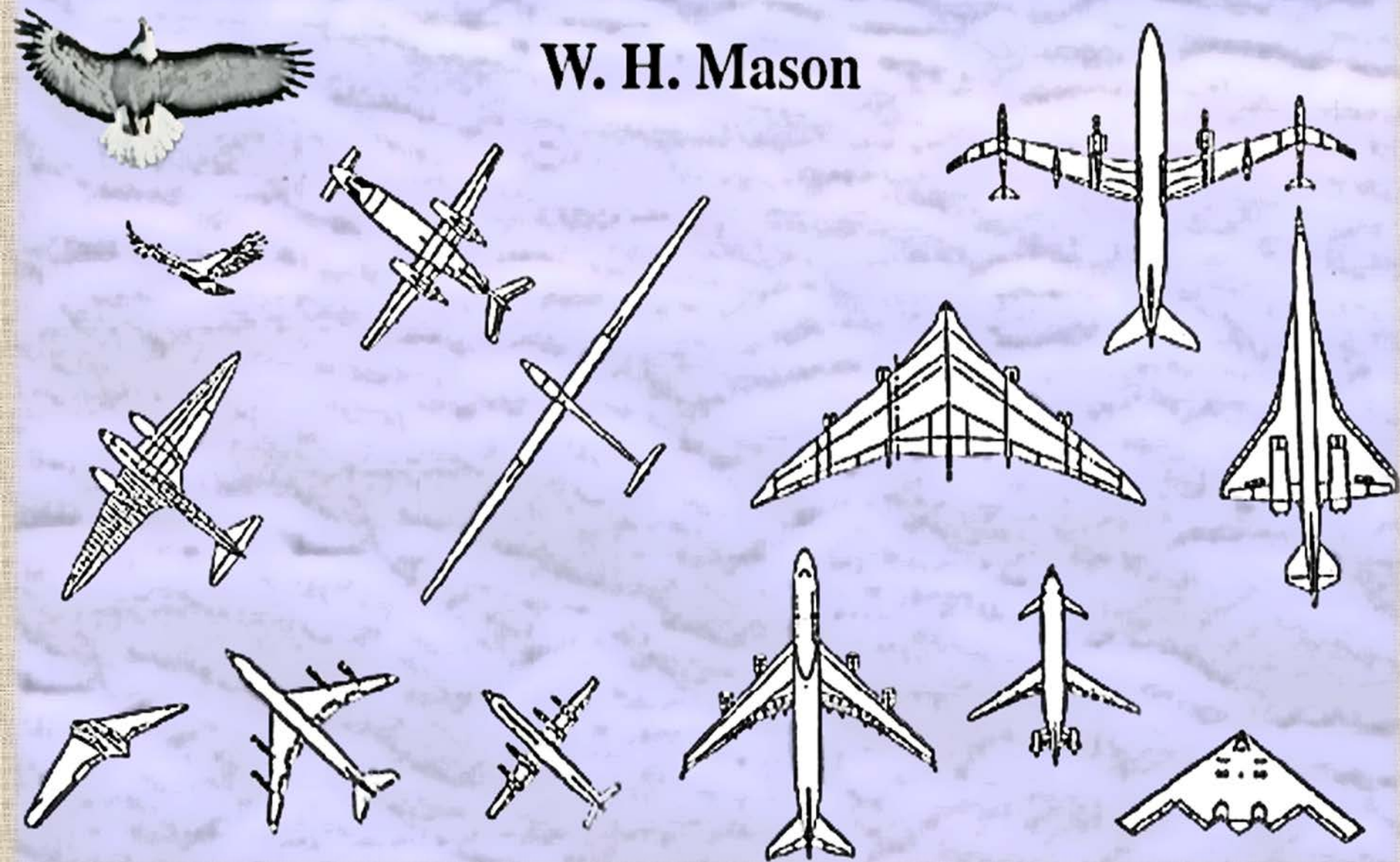
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Why Airplanes Look Like They Do

W. H. Mason



About William *Bill* Mason



<http://www.aoe.vt.edu/people/faculty/whmason.html>



Designer

Technology
advances?

A new capability
someone might
pay to have?

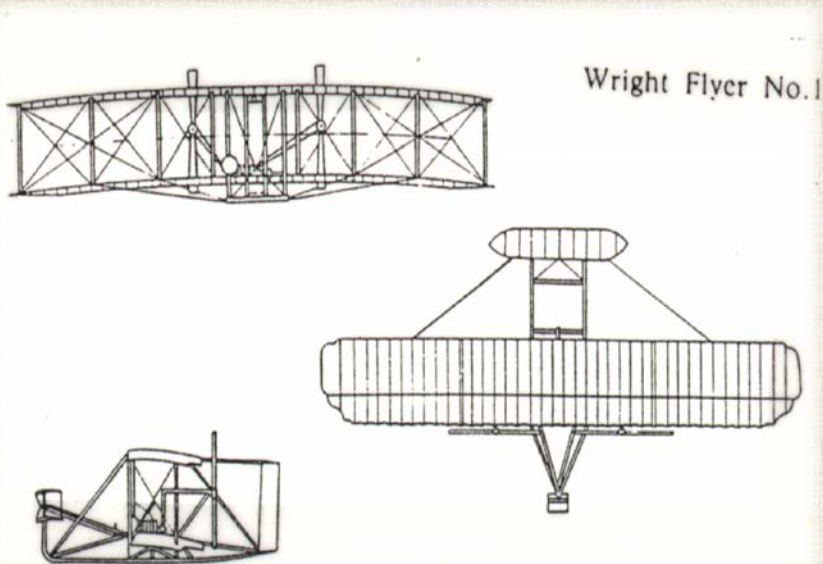
How to exploit
technology for
capability?

Configuration
Concept

Airplane Shapes Have Changed to Exploit Advances in Technology

Configuration Concept:

- Payload
- Lifting surface arrangement
- Control surface(s) location
- Propulsion system selection
- Landing Gear



Wright Brothers:

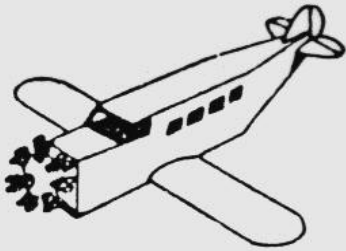
- Innovative control concept
(more important than stability)
- “Light weight” propulsion
- Continual design evolution/refinement

Basic Laws of Airplane Design

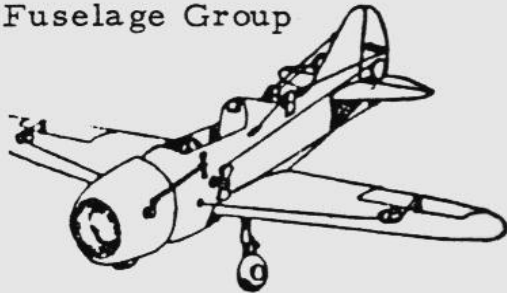
- *Simplicity is the essence of true elegance—
- it can also save weight and/or reduce cost.*
- *If you can't build it, you can't sell it.*

John McMasters
Boeing Commercial Airplane Company

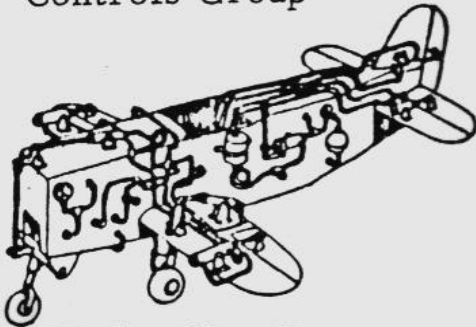
Beauty lies in the Eye of the Beholder



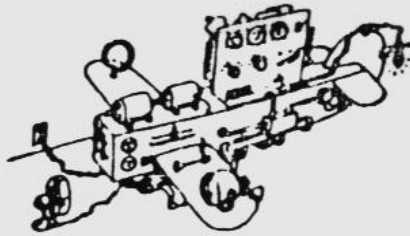
Fuselage Group



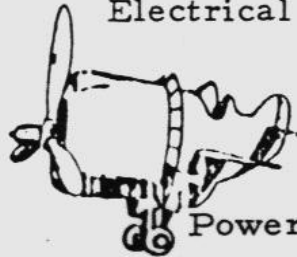
Controls Group



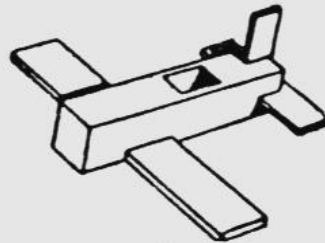
Hydraulics Group



Electrical Group



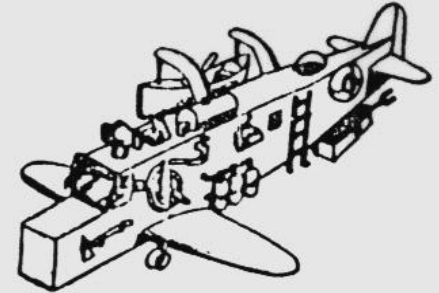
Power Plant Group



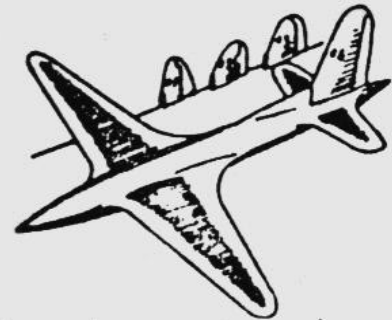
Loft Group



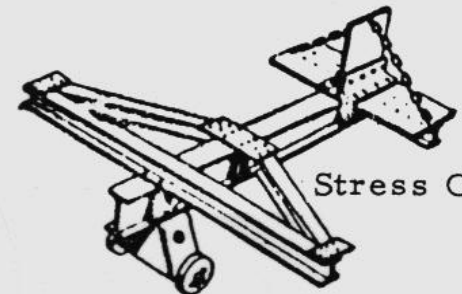
Production Engineering Group



Equipment Group



Aerodynamics Group



Stress Group

“Dream Airplanes” by C.W. Miller,
as shown in *Fundamentals of Aircraft Design*, by L.M. Nicolai

Features of Good Aircraft



- Aerodynamically efficient, including propulsion integration (streamlining!)
- Must balance near stability level for minimum drag
- Landing gear must be located relative to cg to allow rotation at TO
- Adequate control authority must be available throughout flight envelope
- Design to build easily and have low maintenance costs
- Should be quiet, and have low emissions

Key Technologies

- Aerodynamics
- Propulsion
- Structures

in the late 70s:

- Flight controls

in the 80s and early 90s:

- Systems/avionics/observables & Manufacturing

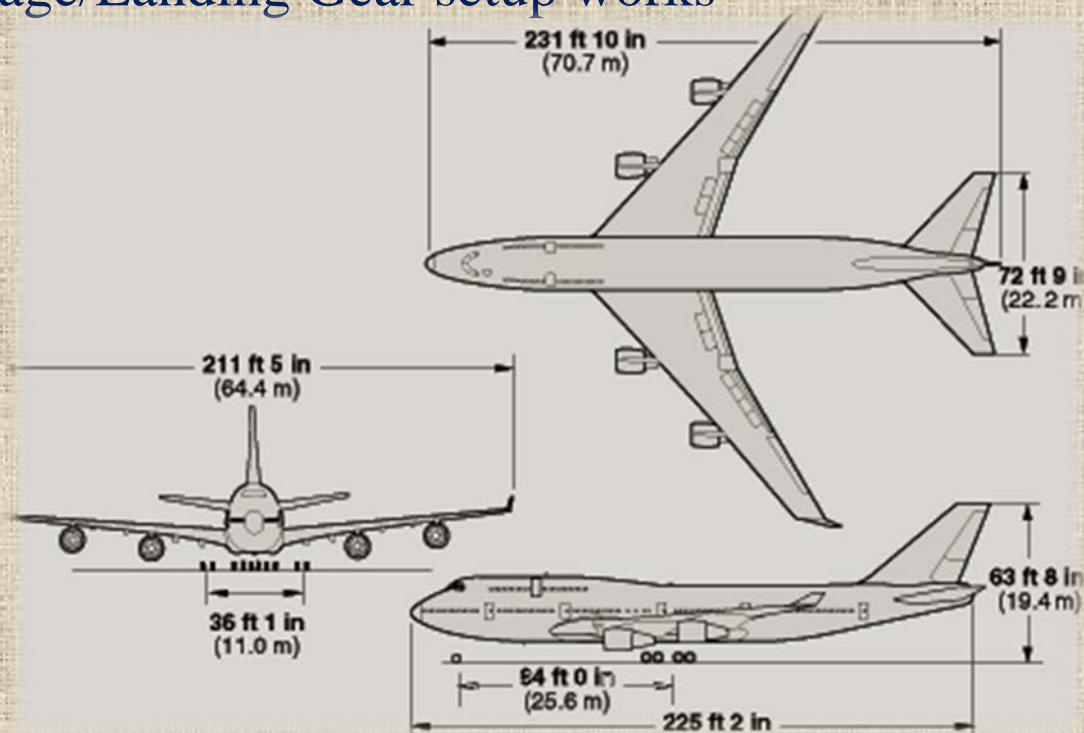
today:

- the design *process* - (includes MDO)

Amazingly Tricky to Integrate Advances in Each Technology

Conventional Subsonic: A Baseline

- Payload distributed around cg
- Longitudinal control power from tail (with moment arm)
- Vertical Tail for directional stability, rudder for control
- Minimum trimmed drag at near neutral stability
- Wing/Fuselage/Landing Gear setup works



Boeing 747-400, source: www.boeing.com

Layout Options

- Where do you put
 - the wings?
 - the engines (in fact, what kind?)
- Where do you put the control surfaces?
 - what options are available?
- Do you have room for the landing gear?
- Possible innovative designs?

A few WHYs ??

- ❑ *Why podded engines below wing ?*
- ❑ *Why Swept wing ?*
- ❑ *Why Forward Sweep ?*
- ❑ *Why Variable sweep ?*
- ❑ *Why Canards ?*
- ❑ *Why Flying Wing ?*
- ❑ *Why Three-surfaces ?*
- ❑ *Why Winglets ?*
- ❑ *Why Thrust Vectoring ?*
- ❑ *Why ? Why ? Why ? Why ? Why ?*

Why Put Engines in Pods on Wing?

- load relief on wing: weight savings
- access to work on engines (maybe)
- safety
- can be low drag

Original idea by the British – in wing!



If it's small, can't put them below wing



Boeing Made Wing Mounted Engines Work



Why Sweep the Wing?



Subsonic (usually small)

- Adjust wing aero center relative to c_g
- On flying wing, get moment arm length for control

Transonic (significant, $30^\circ - 35^\circ$)

- Delay drag rise Mach (compressibility effect)
 - Definition of the Drag Divergence Mach no.?

Supersonic (large, $45^\circ - 70^\circ$)

- Wing concept changes, must distribute load longitudinally as well as laterally
- Reduce cross-sectional area and area variation



Wing sweep increases wing weight for fixed span

Why Sweep the Wing Forward?



X-29



SU-47